TABLE OF CONTENTS

	Page
INTRODUCTION AND SUMMARY	1
RECOMMENDATIONS FOR ADDITIONAL WORK	3
RESULTS	7
Gassing of Organic Materials	7
Rate Controlling Mechanism for Gassing	10
Gassing Rates for Materials in Stabilization and Control System	17
Time-Line Analysis of Gassing Rates for SCS Generation Rate Profile for SCS	17 23
Trace Materials not Generated by Stabilization and Control System	27
Volatile Materials of Biological Origin	31
Skin Secretions Flatus Respiratory Gases Summary	31 46 49 50
Selecting Trace Materials for Use in Trace Material Control Unit	53
Generation Rate Patterns	58
Programming the Trace Material Control Unit	62
Simulation of Depletion Processes	62
Selection of Reasonable Leak Rate and Circulation Rate Through Environmental Control System	64
A TMCU Program for Closed System without Trace Material Sinks	65
Programming the TMCU to Simulate a Closed System with Trace Material Sinks	71
REFERENCES LIBRARY COPY	74

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Figures

		Page
Figure 1.	Relative Rate of Weight Loss and Actual Weight Loss in Oven Test for Various Materials in SCS	12
Figure 2.	Distribution of Oven Weight Loss Data for Various Materials in SCS	13
Figure 3.	Comparative Weight Loss Data for Oven Test (350°F) and Screening Test (200°F)	16
Figure 4.	Vapor Pressure and Evaporation Rate of Water	36
Figure 5.	Program of TMCU to Simulate System Without Trace Material Sinks	70
Figure 6.	Program for TMCU to Simulate Leak on Operating SCS	73
	<u>Tables</u>	
Table 1.	Gassing Rates for Materials in SCS Based on Oven Weight Loss Data	15
Table 2.	Modified Power Profile for SCS	18
Table 3.	Material Inventory of SCS, and Distribution of Materials (in Grams) Among Subsystems	19
Table 4.	Estimated Weights of Materials in SCS at 150°F Versus Mission Time	20
Table 5.	"Temperature Profile" for Materials in SCS	22
Table 6.	Generation Rates of Carbon Monoxide and Total Organics for SCS During Period of 40 to 54 Days	24
Table 7.	Amounts of Distribution of Trace Contaminants from Materials in SCS (14-Day Mission)	25
Table 8.	Trace Contaminants from SCS During 14-Day Mission	26
Table 9.	Amounts of Trace Contaminants Generated by Other Than Electronic Gear (14-Day Mission)	29
Table 10.	Disposition of Biological Sources of Atmospheric Contamination	30

		Page
Table 11.	Volume and Composition of Flatus	47
Table 12.	Production Per Man of Atmospheric Contaminants	51
Table 13.	Trace Contaminants from Biological Sources During 14-Day Mission	52
Table 14.	Characterization of Trace Materials	56
Table 15.	"Relative Toxicity" Ratings of Trace Contaminants	57
Table 16.	Final Generation Rate Pattern for SCS System (14-Day Totals)	59
Table 17.	Final Generation Rate Pattern for Materials, Processes, and Men (14-Day Totals)	61
Table 18.	Calculated Concentrations of Trace Materials	67
Table 19.	Program for TMCU to Simulate System Having No Trace Material Sinks	69

FINAL REPORT

on

TRACE MATERIAL GENERATION RATE SIMULATOR.
TASK B: STUDY OF GENERATION RATE PATTERNS

to

HONEYWELL, INC. AERONAUTICAL DIVISION

from

NORTH STAR RESEARCH AND DEVELOPMENT INSTITUTE
May 28, 1965

INTRODUCTION AND SUMMARY

This document presents the methods and results of studies on tracematerial generation rate patterns anticipated in the habitable area of a
spacecraft on a 14-day mission. The chief objective of the research was to
provide programs for operation of a trace material control unit (TMCU) which
would simulate both actual generation rates and equivalent rates modified to
account for the presence of trace material sinks. A second objective was to
disclose any inadequacies in the inventory of a space capsule, in the identification and characterization of trace materials, and in the kinetics of
gas evolution which should be remedied to permit a more reliable study of
generation rate patterns.

Research methodology and the results obtained by application of the methodology to the problem of estimating generation rate patterns for trace

materials are presented. The trace-material sources are the material, equipment, and the men in a six-cubic-meter capsule designed for a 14-day lunar mission.

The first part of the report is an exposition of the rate-controlling mechanism for the gassing of organic materials. A diffusion-controlled mechanism provides the mathematical model for calculation of gassing rates versus time. It is applied to the components of the stabilization and control system (SCS) to arrive at trace-material generation rates. Correlation of the findings with data on analyzed atmospheres after Mercury flights led to an estimate of the total amount of electronic gear, and to estimates of the kind and rate of generation of other trace materials not associated with electronic gear. Volatile materials of biological origin were added to the listing. Results of all calculations are provided in tabular form in the report.

A scheme was devised to use available numerical data on the toxicity of materials to obtain "relative toxicity" ratings. Some trace materials could then be substituted for others to arrive at a smaller list of materials for use in the TMCU. The "relative toxicity" ratings have been tabulated.

The last part of the report presents a methodology for simulating the action of trace-material sinks in order to develop programs for the TMCU. Programs designed to simulate closed systems with and without trace-material sinks require the actual use of open loops. The programs are, in general, within the design capabilities of the trace-material control unit.

RECOMMENDATIONS FOR ADDITIONAL WORK

The current analysis was undertaken with the object of arriving at trace material generation rates based on available information. It was, and still is, apparent that data are lacking for complete and accurate development of generation rates, and for the making of meaningful toxicity predictions.

To gather the requisite data on generation rates, three major alternatives may be considered. These are:

- Analysis of the accumulation of gaseous contaminants in a space capsule carried through a simulation of actual use.
- 2. Development of understanding based on mathematical and experimental analyses of the actual processes contributing to gaseous contamination occurring in each component material.
- Development of empirical data on gaseous output from each component material.

The first alternative is surely the easiest, but at the same time the least satisfactory. It does not provide the basic data which will permit prediction of effects on gaseous contamination when alterations are made in materials, components, or operating conditions.

The second alternative is ideal but may or may not be practical. A first step would be to examine its practicality. We visualize that its pursuit could progress through the following phases:

- 1. Mathematical treatment of one or two cases in which assumptions are made as to the existance of several concurrent processes (oxidation, thermal degradation, and diffusion). Such an analysis would serve to examine the ease with which a detailed understanding of the processes involved could be put to use, if available. It would also serve to identify the important variables to which attention should be given in gathering and using experimental data.
- 2. A complete inventory of all materials, equipment, and processes in Apollo C/M.
- 3. Forecasts of the expected temperature profile as a function of time for each material in the inventory. This is necessary as a guide to testing procedures and as part of the information needed to predict gassing rates.
- 4. Information on weight, surface area, and volume of each item of the inventory together with information on proximity to diffusion barriers which will restrict access of outgassed material to the atmosphere.
- 5. Measurements of outgassing designed to give information on the processes involved and their rates. This would mean that data should be gathered at a series of temperatures under conditions such that diffusion rates are not a limiting factor. This could be accomplished by making measurements of outgassing of very thin sheets or of powders. Measurements should be made in the presence and in the absence of oxygen to distinguish between thermal degradation and oxidative processes. By repeating studies using thicker pieces of known volume, diffusion rates could be calculated.

Such an approach would give basic data which would be interpolated and extrapolated with confidence as alterations in components are made (providing the same materials are used). If the labor of such an approach

appears overwhelming after completion of step one, or even before its completion, then the third alternative approach should be undertaken. This approach would require:

- A complete inventory of all materials, equipment, and processes in Apollo C/M.
- 2. Forecasts of the expected temperature profile as a function of time for each material in the inventory. This is necessary as a guide to testing procedures and as part of the information needed to predict gassing rates.
- 3. Studies of outgassing of actual components of suitable samples of actual components. Such studies would probably provide sufficient data for predictive purposes if it included measurements of gases produced at three different time intervals at each of three temperatures. In actuality, it probably would not be necessary to carry out studies on each component. Initial probes would undoubtedly show the gas contribution from some classes of materials to be negligible.

This third alternative would not be capable of as great extrapolation as the second. From the practical standpoint, it may be more attractive, and could be undertaken instead of the second alternative.

Contaminants arising from biological sources similarly require additional attention before their identity and quantity can be predicted with confidence. The following units of work are recommended:

1. A literature search in depth for additional information on the identity and generation rates of volatile materials from biological sources. The estimates of these materials set forth in this report have been based largely on secondary sources. A study in which original sources could be studied and compared in detail would yield more meaningful estimates. Interviews with investigators who are actively concerning themselves with such questions as the identity of the components of sweat, the effect of bacterial action on sweat, and the identity of flatus components would be particularly helpful in sharpening and evaluating the estimates.

- 2. An experimental study of expired air to identify and measure normal trace contaminants. Because of the large volume of respiratory air, the presence of very small amounts of foreign chemical species could make a large contribution to the contamination of the space-capsule atmosphere. Very little evidence of systematic investigation of this source was evident from our exploration of the literature. Such studies would seem to merit high priority. Some ideas on the necessary refinement of experimental approaches have been conceived.
- 3. A study to identify and measure the unidentified components of flatus. Dr. Murphy of the United States Department of Agriculture had estimated that as much as one percent of flatus components remain unidentified. This could be an important source of contamination and is deserving of continued research emphasis.
- 4. A study of apocrine sweat. In work on sweat constituents little attention has been given to apocrine sweat. Systematic studies of its constitution are needed.

These comments relate to the identification and quantification of those materials which might predictably appear as contaminants of the space capsule atmosphere. The assessment of their significance in terms of effects on man represents an area where very little meaningful work is available. This is especially true when one considers possible additive and synergistic effects and when one considers possible effects on behavior and efficiency. Studies in which the effects of combinations of the significant contaminants on behavior as well as upon the usual indicators of toxicity are called for. The initial experiments should be in animals with final confirmation of safety in man.

RESULTS

Gassing of Organic Materials

When the partial pressure of trace materials in the atmosphere is low, the rate-controlling factor in the degassing rate is the mechanism by which these volatiles are released. This mechanism is not unique. For saturated systems, there is a constant "drying" rate corresponding to evaporation as from a liquid of constant area. This is followed by a falling rate over a period of time during which the evaporation rate is proportional to the fraction of the surface which is wet. Finally, there is another period of falling rate during which liquid evaporates from the surface as fast as it can get there by diffusion from the interior of the solid.

Materials acceptable for space-travel missions will gas according to some modification of the last mechanism. This follows because baking of coatings and "running in" of equipment prevents the attainment of saturation of the surface with volatile materials. The modification of simple diffusion arises from slow, but continuous, generation of the same or new contaminants by thermal degradation and oxidation. These fresh trace materials must also be lost by a diffusion process, except for a small fractional part generated at the surface. Thus, the over-all rate of generation of trace materials will be diffusion-controlled.

Diffusion-controlled gassing will react to temperature changes by the obvious change in magnitude of the diffusion constant and by changes in

rate of production of trace materials by oxidation and thermal degradation. For a given temperature the trace contaminants will be lost according to a particular solution of Fick's law of diffusion for nonstationary state of flow. The solution used for this program⁽¹⁾ is adapted to the empirical gassing data available and to the geometry of the test materials. The concept used is that of drying a slab of material from two sides. It is, of course, only an approximation in view of the variable geometry of test samples and of materials used in a spacecraft. However, it will define the general exponential decline with time of trace contaminant content of a material:

$$\log\left(\frac{Q_t}{Q_0}\right) = \log\left(\frac{3}{\pi^2}\right) - \frac{\pi^2 Dt}{9.2a^2}$$

where Q_t = average concentration remaining at time t

 Q_0 = initial uniform concentration

a = thickness

D = diffusion constant

t = time

Use of the above equation serves two purposes. First, the gassing process can be studied as a diffusion-controlled process for materials on which gassing data are available. Second, gassing rates can be estimated for specific time intervals and for different temperatures. The accuracy of

estimation will be unknown in the case of samples of varying geometry undergoing a wide variety of decomposition reactions. A more accurate procedure could be followed in a kinetic study of one material.

The first objective -- showing that all materials gas according to a diffusion-controlled mechanism -- was met by showing: (1) that only random deviations exist in the relative rates of outgassing, and (2) that all the relative rates of outgassing conform to the pattern of equation [1]. The term "relative rate" here permits use of data based on samples of different materials with different Q_0 values and different geometries. The second objective -- estimating gassing rates for different temperatures over the time interval of the Apollo mission -- was met by using a reasonable energy of activation for diffusion, 6000 cal per mole, and by assuming no change in geometry between test samples and materials in the spacecraft. It is apparent that experimental kinetic studies are required to determine the temperature coefficients of specific reactions of thermal degradation and oxidation, and their contributions to changes in gassing rates with temperature.

Rate Controlling Mechanism for Gassing

Oven weight-loss data versus time were used to examine the random variation of relative rates of weight loss among all the materials studied in Honeywell's program⁽²⁾ of gassing testing for Apollo C/M Stabilization and Control System. The measurable weight losses for seven days at 350°F ranged from 0.1 to 68 percent by weight. The relative rate of weight loss between one and seven days was defined as the difference between the logarithms of percentage weight loss at these two times. Plotted against percentage weight loss after seven days in Figure 1, the relative rates show a random pattern. This suggests that the variations are due to experimental error rather than to a given mechanism. There is no evidence for more than one rate-controlling process, and the data can be represented as well by a single process, say diffusion, as by several.

To demonstrate this point further, the same relative rates were plotted on probability-log scales in Figure 2. The cumulative percentage of samples refers to those samples reported in the earlier testing program⁽²⁾. Note that the random nature of data in Figure 1 is confirmed by the organized pattern in Figure 2. Moreover, the relative gassing rates for different samples of one generic material (an epoxy designated as 6020) are scattered over the entire line in Figure 2. Since it is reasonable to assign only

one rate-controlling mechanism to one generic material, and since the scatter of data is no greater for the several materials than for the one material, it therefore, seems appropriate to assign only one rate-controlling mechanism to gassing of all the materials tested.

Probably the most obvious rate-controlling mechanism to consider is diffusion. Accordingly, the data on oven gassing were normalized by treatment according to equation [1]. This was aided by a redefinition of terms:

Q_o = extrapolated value for weight loss in percent at infinite time,

 Q_{t} = weight loss in percent at time, t. The following treatment was applied then to the oven test data for the more limited number of organic materials in the Stabilization and Control System.

By a series of approximations, a Q_O value was selected for each material such that plots of log (Q_t/Q_O) versus time intercepted the log axis near $8/\pi^2$ as defined by equation [1]. All data conformed reasonably well to this treatment. The results supplied a bonus by indicating which materials were undergoing almost total degradation; for example, one material had a Q_O value of 91 percent.

The chief use made of equation [1] was to calculate weight loss for any time interval. This was critical to the program because time zero for a space mission is not time zero for manufacture of a plastic or coating of

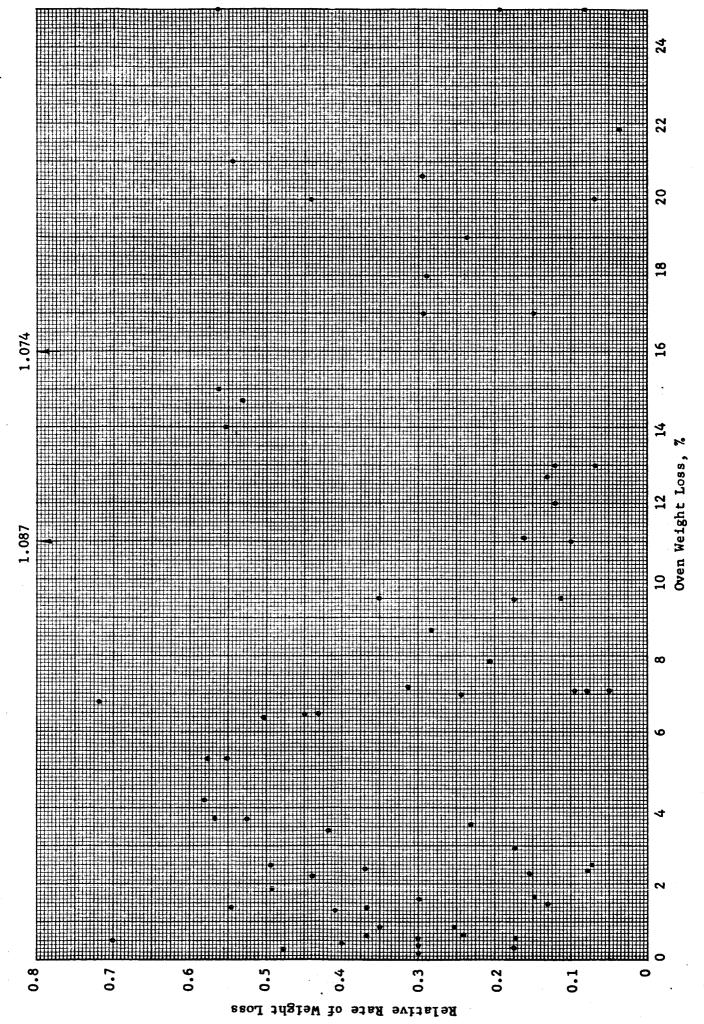


FIGURE 1 RELATIVE RATE OF WEIGHT LOSS AND ACTUAL WEIGHT LOSS IN OVEN TEST FOR VARIOUS MATERIALS IN SCS

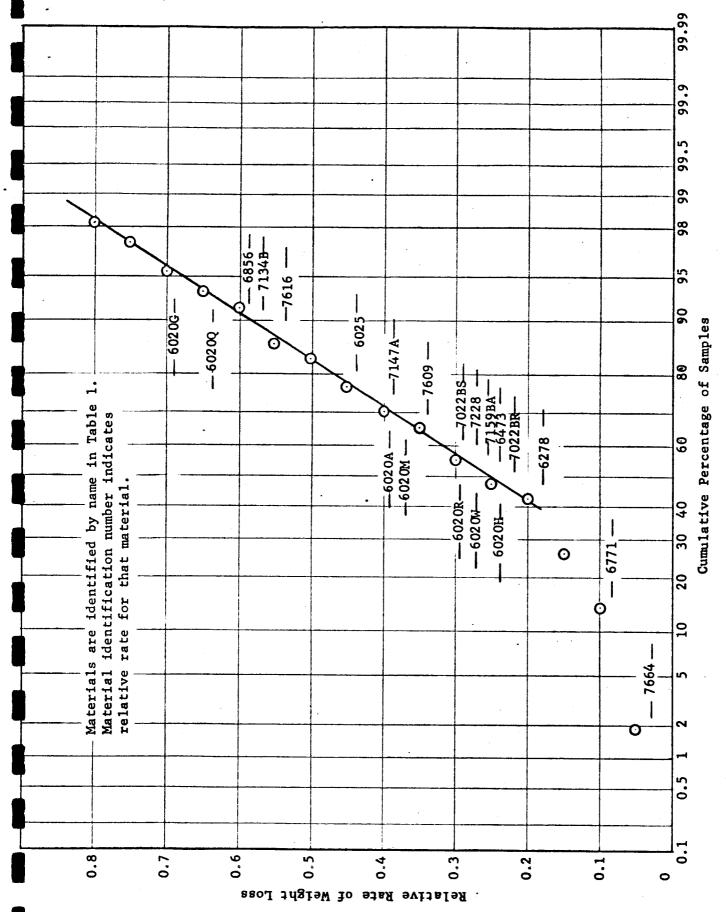


FIGURE 2 DISTRIBUTION OF OVEN WEIGHT LOSS DATA FOR VARIOUS MATERIALS IN SCS

an electrical component. Rather, it is the beginning of a 14-day period following approximately 1000 hours of operation of each system during mortality, running-in, and check-out tests. The period selected for this study was the period between 40 and 54 days after manufacture.

Table 1 contains the gassing rates calculated from oven weight-loss tests at 350°F. These rates are reported in day -1, i.e. the slope of equation [1]. The table also includes calculated rates for 150°F based on an energy of activation of 6000 cal per mole. This temperature coefficient is significant providing there is no change in the geometry of the materials; with no geometry change, the diffusion constant is the only temperature-sensitive term in the slope of equation [1]. One recognizes but does not account for the higher temperature coefficient which is characteristic of the inherent processes of oxidation and thermal degradation.

It is difficult to estimate the reliability of the temperature correction. The selected energy of activation is reasonable for a diffusion process, however, and there were no experimental data at low temperature from which an energy of activation could be calculated. There are, in fact, gassing data for near $200^{\circ}F$ based upon single points in time -- two days usually. The relationship between weight loss during seven days at $350^{\circ}F$ and two days at $200^{\circ}F$ is so tenuous (Figure 3), that calculations based on oven test data at $350^{\circ}F$ and the assumed $\triangle E$ are preferred over use of the gassing data at $200^{\circ}F$.

TABLE 1

GASSING RATES FOR MATERIALS IN SCS BASED ON OVEN WEIGHT LOSS DATA*

Total Gassing Rates at 150°F Between 40 and 54 Days (Percent)	0.075 1.42 0.074 45.8 0.276 0.004 1.63 0.778 0.074 0.567 0.407	0.024
, Day-1	0.2 × 10 ⁻² 0.275 × 10 ⁻² 0.25 × 10 ⁻² 0.039 × 10 ⁻² 0.825 × 10 ⁻² 0.7 × 10 ⁻² 0.29 × 10 ⁻² 0.426 × 10 ⁻² 0.92 × 10 ⁻² 0.92 × 10 ⁻²	0.300 × 10 ×
$n = \frac{H^2 Dt}{9.2 a^2}, Day^{-1}$	4 4 4 4 3 3	- OT X 7/.7
Intercept	1.8 0.80 24 0.90 1.55 0.78 Estimated from gassing 49 0.77 0.81 7.5 0.95 0.75 0.95 0.75 0.95 0.75 0.95 0.81 7.0 0.93 Estimated from gassing 0.85 Estimated from gassing 0.85 Estimated from gassing 0.85 Estimated from gassing	0.00.0
Qo, Percent	1.8 24 1.55 Estimated 3.1 90.77 Estimated 4.9 7.0 4.4 Estimated 0.85 Estimated 0.85	74.0
Spec, No.	6473 71348 6762 7147A 7664 7022 60206 7228 7616 6025 6733 7570	401
Material Name	Glass epoxy laminate Glass epoxy varnish Fiberglass tape Lubricant Silicone rubber Polyester enamel Epoxy glass laminate Nylon tying cord Polyester tape Epoxy adhesive Urethane foam Acrylic Rubber adhesive Epoxy foam Acrylic enamel	Vicon A

*Note that where oven test data were erratic or not available, estimates were based on measured weight loss in gassing-test data for $200^{\circ}F$.

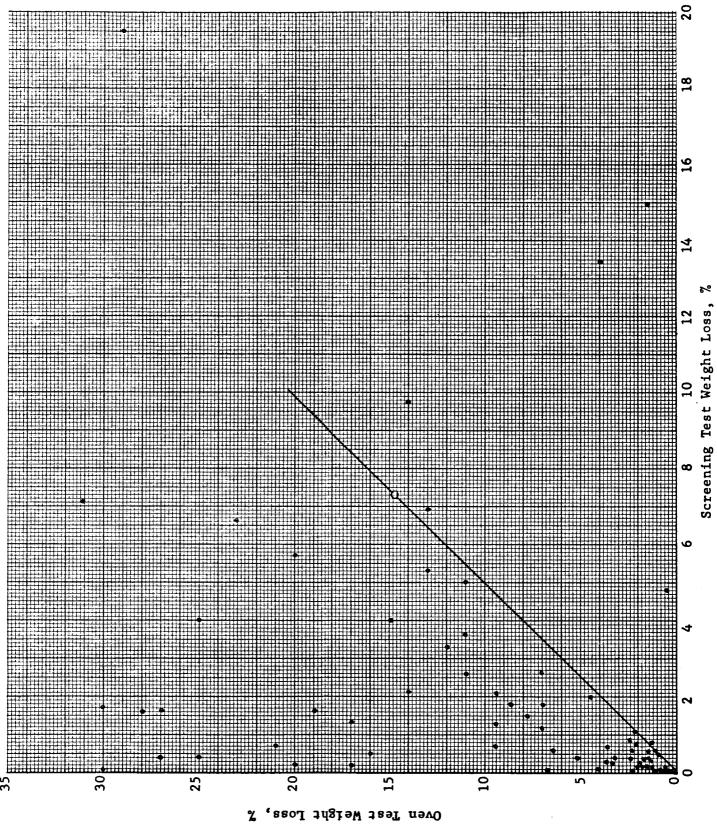


FIGURE 3 COMPARATIVE WEIGHT LOSS DATA FOR OVEN TEST (350°F) AND SCREENING TEST (200°F)

Gassing Rates for Materials in Stabilization and Control System

Time-Line Analysis of Gassing Rates for SCS

A time-line analysis of gassing rates for the SCS was based on a power profile for its subsystems and a listing of organic material weights by subsystem. A modified power profile, Table 2, presents the fraction of full power at which each of the 13 subsystems operates at any time during a 14-day lunar mission. This table includes some arbitrary selections of maximum power, and some "smoothing" of power inputs by a time-weighted average where the power profiles were jagged. The abbreviated profile is more than adequate in consideration of the modest fluctuation in power level with time.

The abbreviated power profile was used to project material temperatures. All organic materials within a subsystem were arbitrarily considered to be at 150°F whenever that subsystem was operating at maximum power, or at 80°F when that subsystem was operating at zero power. For intermediate power levels, it was assumed that the weight of material at 150°F is proportional to the power level expressed as fraction of full power for the subsystem concerned. Accordingly, the material inventory of the SCS was distributed among the subsystems (Table 3) and used to calculate weights of material at 150°F in each subsystem at any time (Table 4). The balance of the weight of each material could be considered to be at 80°F. For ease of calculation,

TABLE 2

MODIFIED POWER PROFILE FOR SCS*

			avg.															-						
	13	1.0	0.12 gr		ò	0.00	1.0	0.00	1.0	0.0	1.0	0.0	1.0	0.0	1.0	0.0	1.0	90.0		0.014	0.3		1.0	1.0
	12	1.0				1								-				·						→
	11	1.0	•••																					→
	임	0.21	69 0			1.0		0.62	1.0	0.62	1.0	0.62	1.0	0.62	1.0	0.62	1.0	0.62					;	0.62
	6	0.0																					-	→
Ħ	ω	1.0		· · · · ·											J. 5									→
Subsystem	7	1.0					-	_																→
	9	0.0				<u></u>																		→
	2	1.0		- 											-					-				→
	4	1.0				······································																-		→
-	m	1.0																						→
	2	1.0	<u>-</u>										-			-)
		1.0	0.725	1.0	0.725		1.0	0.725	1.0	0.725	1.0	0.725	1.0	0.725	1.0	0.725	1.0	0.725	1.0	0.725	1.0	0.725	1.0	1.0
Time	Hours	H 9	12	45	46	51 79	80	86	112	120	130	131	166	172	178	182	246	250	267	268	326	327	333	336

*Entries are fractions of maximum power level for each subsystem.

TABLE 3

MATERIAL INVENTORY OF SCS, AND DISTRIBUTION OF MATERIALS (IN GRAMS) AMONG SUBSYSTEMS

Spec. No.	٥		2	3	4	5	9	Subsystem 7	е 8	6	10	1	. 12	13	Tota1
					-		,		,		*	**	*	7	757 (7
6473 299 1		-	_	126.1				228.4	63.7				12.24	17.5	747
7134B 53.4 32.9		32.9			37.5		29.8	33.6						0.69	188
6//1 6762						0.48	6.0				. /	9.92	9.84	6.57 2.25	6.6 14.4
7147A 54.1 83.3		83.3			82.5			80.5						1.14	302
7664							17.5		14.6	8.9	12.2		9.73	5.59	4.99
7560 260.4	260.4	260.4			228.4	•					110	345.6			776
7022 6856						36.1				17.8	26.8		4.15		67
6020G 915.3	915.3	915.3) ,)	10.73			926
7228						7						15.09			15.1
7616												34.14			34.1
												0.33		-	0.33
6733 95.7 93.8		93.8			71.7	1.9		72.8				2.83			339
7609 1383.8	1383.8				830.3			869.1							3870
7570					12.4			,							12.4
7765 0.22	0.22	0.22													44
7541	-	_					¢			4.2					4.2
/159							0.21								17.0

TABLE 4

ESTIMATED WEIGHTS OF MATERIALS IN SCS AT 150°F VERSUS MISSION TIME

	7159	0	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	± .	=	=	=	=	=	=	=
	7541	0	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	<u>:</u>	=	=	=	=
	7765	0.44	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=
	7570	12.4	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=
	7609	3870	3870	3500	3500	3870	3500	3500	3500	3870	3500	3870	3500	3870	3500	3870	3500	3870	3500	3870	3500	3870	3500	3870	3500	3870	3870
	6733	340	340	312	312	340	312	312	312	340	312	340	312	340	312	340	312	340	312	340	312	340	312	340	312	340	340
	6025	0.33	=	=	=	E	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=
	<u>7616</u>	34.1	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=
	7228	15.1	= .	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=
Material	6020G	926	=	=	=	=	=	=	=	=	=	=	=	= .	=	=	=	=	=	=	=	=	=	=	=	=	=
Ma	<u>6856</u>	16.4	=	=	=	=	=	=	=	=	= .	=	=	=	=	=	, =	=	=	=	=	=	=	=	=	=	=
	7022	5.	ij	ij	9	9	56.9	9	ä	1:	•		•		•			7	9		9	9	•	9	9	9	•
	7560	857	921	921	306	902	905	902	546	944	905	544	902	544	902	544	905	546	902	544	902	902	905	905	902	905	905
	7664	32.5	39.5	39.5	32.6	32.6	32.6	32.2	36.8	42.1	31.9	42.1	31.9	42.1	31.9	42.1	31.9	42.1	31.9	42.1	32.2	32.2	32	33.6	33.6	37.5	37.5
	7147A	302	302	287	287	301	286	286	286	302	286	302	286	302	286	302	286	302	286	302	286	286	286	302	. 586	302	302
	6762	13.5	13.5	13.5	11.5	11.5	11.2	11.2	11.2	13.5	11.2	13.5	11.2	13.5	11.2	13.5	11.2	13.5	11.2	13.5	11.4	11.4	11.2	11.5	11.5	13.5	13.5
	<u>6771</u>	9.9	9.9	9.9	0.7	0.7	0.7	9.0	4.0	9.9	0.00	9.9	0.00	9.9	0.00	9.9	0.00	9.9	0.0	9.9	9. 0	4.0	0.1	2.0	2.0	9.9	9 • 9
	7134B	158	158	143	143	157	143	143	143	158	143	158	143	158	143	158	143	158	143	158	143	157	143	157	143	158	158
	6473	747	747	665	650	732	650	648	648	747	648	747	648	147	648	747	849	747	849	747	649	731	849	735	653	747	747
Time,	Hours	-	9	12	13	45	46	51	79	80	86	112	120	130	131	166	172	178	182	246	250	267	268	326	327	333	336

these latter weights were converted by a factor of 8.15 to corresponding weights which would gas at the same rate at 150°F in order to obtain the final temperature profile of Table 5.

From Table 5, generation rates may be correlated with individual materials or with the total system. It should be noted that the time-temperature profile indicates only small fluctuations of generation rate with time. The time-weighted averages of the weights of each material at 150°F can be used to estimate trace material generation rates.

TABLE 5

"TEMPERATURE PROFILE" FOR MATERIALS IN SCS, GRAMS AT 150°F*

	7159	1.6	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	7. E	=	=	=	=	=			۲.۵	
	7541	0.515	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	E			0.515	
	7765	0.44	=	=	=	=	=	=	=	=	=	=	=	=	=	=	z	E	=	=	=	=	=	=	=	=	=		;	4.	
	7570	12.4	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=			12.4	
	7609	3870	3870	3545	3545	3870	3545	3545	3545	3870	3545	3870	3545	3870	3545	3870	3545	3870	3545	3870	3545	3870	3545	3870	3545		3870		6	3590	
	6733	340	340	31.5	315	340	315	315	31.5	340	315	340	315	340	315	340	315	340	315	340	315	340	315	340	315	340	340			318.5	
	6025	0.33	=	=	=	=	=	=	=	=	=	=	=	=	=	=	z	=	=	=	=	=	=	=	·. =	=	=			0.33	
	7616	34.1	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=		;	34.1	
	7228	15.1	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=	=,		1	15.1	
Material	6020G	926	=	=	=	=	· =	=	=	=	=	=	=	= .	=	=	=	=	=	=	=	=	=	=	=	=	=		,	926	
Mat	6856	œ.	18.6	8	œ	=	=	=	=	=	=	.=	=	=	=	=	, =	=	=	=	=	=	=	=	=	=	=		,	18.6	
	7022	48.5	62.4	62.4	58.1	58.1	58.1	58.1	62.4	62.4	58.1	7	∞	7	∞	~	58.1	7	58.1		œ	58.1	58.1	58.1	58.1	58.1	58.1		1	58.7	
	7560	898	924	924	907	206	907	907	944	944	206	944	206	944	206	576	907	576	907	944	905	905	905	905	905	905	206		,	910	
	7664	36.7	42.8		36.8	•	•		•	•	36.1	•	•	45.1	36.1	45.1	36.1	45.1	•	•	•	36.4	36.2	•	37.6	41.1	41.1		,	37.3	
	7147A	302	302	683	68;	101	88	88	88	302	88	302	88	302	88	302	88	302	88	302	88	288	883	302	883	302	302			290	
	6762	9.	9	9.	11.9	0	ဖ	9	9	9	9	3.6	1.6	3.6	1.6	3.6	1.6	9	1.6	3.6	œ	1.8	9.	0.	6	9.	9.			11.9	
	6771 (9.	9.	9.	ຕຸ	ຕຸ	ຕຸ	.45	.45	9.	ထ	9	ω.	9	80	9	ω,	9	ω.	9	.45	0.45	•	•		•	9.6	٠	;	1.64	
	7134B																					161								151	
	6473 7																					733 1								673 1	
Je.	n ml	1 7	7. 9	2	ě e	5 7															٠							me -		Average 6	
Time	E C	*, *	_	17		4			7	ಹ	ã	11:	12(13(13	16	17:	17	18	241	25(267	7 97	32(32	33	336	Time	¥.	Av	•

^{*}Entries are reported as fictitious weights in grams of each material considered as being at 150°F for purposes of calcu-lating gassing rates.

Generation Rate Profile for SCS

Equation [1] was employed to extrapolate weight loss data on the SCS to an operational period of 14 days; a period of 1000 hours, assumed as an average for operational running of all systems before launch (Table 1), was taken into account. The extrapolated weight losses were apportioned between carbon monoxide and total organics for each material (Table 6) using gas identification data from Honeywell's report to North American Aviation (NAA)⁽²⁾. The total organic output of each material was further apportioned among the specific identified and unknown compounds according to relative concentrations of organic contaminants as found in the same study for NAA (Table 7). Trace contaminants were summed over all materials in the inventory to provide a listing of amounts of 12 specific organic compounds and of a group of unknowns (Table 8).

TABLE 6

GENERATION RATES OF CARBON MONOXIDE AND TOTAL ORGANICS FOR SCS DURING PERIOD OF 40 TO 54 DAYS

		Weight		Quantities of	Quantities of Trace Materials	
		Material,*	, 00	00	Total Ogg.	Total Org.
Material	Specification	8	g × 10	Micromoles	8 x 10 ⁵	Micromoles
Glass Epoxy Laminate	6473	673.		162.	35.7	496.
Epoxy Varnish	7134B	151.		84.3	137.	1905.
Fiberglass Tape	6771	1.64		0.18	0.030	0.42
Lubricant	6762	11.9		0.0	3.0 est.**	<42. est.**
Silicone Rubber	7147A	290.		0.0	6.7	93.4
Polyester Enamel	1664	37.3		38.4	2.9	9.04
Epoxy Glass Laminate	7560	910.		(140.	(2.3	(320.
Nylon Tying Cord	7022	58.7		0.0	51.7	718.
Polyester Tape	6856	18.6	1.45	51.8	11.1	155.
Epoxy Adhesive	6020G	926.	٠.	680. est.	10. est.	142. est.
Silicone Adhesive	7228	15.1		0.0	1.4	19.6
Urethane Foam	7616	34.1		0.0	12.0	167.
Acrylic	6025	0.33		0.253	0.039	0.54
Rubber Adhesive	6733	318.5		0.0 est.	(260. est.	(3600. est.
Epoxy Foam	7609	3590.		4307.	565. est.	7850. est.
Photographic Film	7570	12.4	0.0	0.0	ند	<1.0 est.
Marking Ink	7765	0.44	0.0 est.	0.0 est.		0.0 est.
Acrylic Enamel	7541	0.51	<0.004 est.	(0.05 est.		<51. est.
Viton A	7159	1.6	0.021	0.75		\3.7

^{*} These are actual weights corrected on the basis of the power profile for the mission to equivalent weights at 150°F.

^{**}Estimated values arise where accuracy of data are in question or when data are based on gassing results at $200^{\circ}F$. rather than on oven test data at $350^{\circ}F$.

AMOUNTS AND DISTRIBUTION OF TRACE CONTAM

	co.	Total Organic								Ame	ount	of Orga
<u>Material</u>	Micromoles	Micromoles		AC		MEK		EAL	M	IBK	7	rol
6473	162.1	496	60	(0.12)	50	(0.1)	100	(0.2)	100	(0.2)	20	(0.04)
7134B	84.3	1 9 05	419	(0.22)		•		, ,	724	(0.38)		
6771	0.2	0.4	0.25	(0.63)						•		
6762	0	<42										
7 147	0	93.4					93	(1.0)				
7664	38.4	40.6										
7560	<140	<320									105	(0.33)
7022	0	718										
6856	51.8	155	93	(0.6)								
6020G	680	142										
7228	0	19.6					10	(0.56)				
7616	0	167	137	(1.0)								
6025	0.25	0.5										
6733	0	< 36 00	2400	(0.67)	40	(0.01)			220	(0.06)	540	(0.15)
7609	4307	7850										1
7570	0	<1.0	0.14	(0.14)					0.09	(0.09)		
7541	<0.05	⟨51									7	(0.14)
7159	⟨0.75	<3. 7										
TOTALS	5465	15,555	3109		90		203		1044		672	

BLE 7

NANTS FROM MATERIALS IN SCS (14-DAY MISSION)

nic	Trace	Contaminants	, Micromoles	(Mole	Fraction)

	EBE	<u> </u>	YL	N	BAL	1	MC	IPA	AL		HEP	CUM	U	NK
95	(0.05)	629	(0.33)	125	(0.25)	40	(0.08)	0.15	(0.37)				38	(0.02)
		14	(0.35)	9	(0.21)							105 (0.33)	18 110 718 62	(0.44) (0.34) (1.0) (0.4)
													10	(0.44)
		4.0	(0.01)			0.5	(1.0)	1/0	(0.0/)				200	(0.0()
		40	(0.01)						(0.04) (0.94)	480	(0.06)		220	(0.06)
7	(0.16)		(0.09)	0.06	(0.06)				, ,		•		0.62	(0.62)
,	(0.14)	37	(0.72)										3.7	(1.0)
102		720		134		40		7510		480		105	1180	



TABLE 8

TRACE CONTAMINANTS FROM SCS DURING 14-DAY MISSION

Trace Material	Symbol	Amount <u>Micromoles</u>
i-Propyl Alcohol	IPAL	7,510
Carbon Monoxide	CO	5,465
*	AC	3,109
Methyl-n-Butyl Ketone	MIBK	1,044
o-m- or p-Xylene	XYL	720
Toluene	TOL	672
Heptane	HEP	480
Ethyl Alcohol	EAL	203
n-Butyl Alcohol	NBAL	134
Cumene	CUM	105
Ethyl Benzene	EBE	102
Methyl Ethyl Ketone	MEK	90
Methyl Methacrylate	MMC	40
Unknowns	UNK	1,180

^{*}Acetone, ethyl formate and/or methyl acetate.

Trace Materials not Generated by Stabilization and Control System

Honeywell's thorough inventory of the SCS permitted a detailed study of its trace material generation rate profile. No inventory was available for other systems. It was proposed, therefore, to estimate the total generation rate profile for all materials and equipment by some type of extrapolation of data for the SCS. Studies of the atmosphere of Mercury capsules provide the clue for useful extrapolation.

Some of the trace materials known to arise from the SCS were found in Mercury capsule atmospheres. New trace materials found in Mercury capsule atmospheres, but not generated by materials in the SCS, are assumed to arise from other systems than the SCS. Accordingly, it was possible to develop proportionality factors for trace material generation rates in the SCS to account for the contributions of other materials and equipment.

To illustrate the above reasoning, consider the fairly constant proportionality among concentrations of the same materials found in both Apollo SCS and Mercury capsule atmospheres.

	Average Concentration Ratio		
Material Pair	Mercury	Apollo SCS	
TOL/EAL	7 . 55	3.3	
EAL/MEK	3	2.3	
TOL/NBAL	5.75	4.9	

One can presume that equipment on Apollo C/M will introduce trace materials not found in the SCS roughly in proportions found for Mercury atmospheres.

The new materials always found in Mercury atmospheres are Freon 114, vinylidene chloride, methylene chloride, and benzene. Ethylene dichloride was found sometimes, and should be considered here because of its high toxicity and the likelihood of its being present in substantial amount. The relative amounts of these new materials and of one material (ethyl alcohol) common to both atmospheres are as follows:

Trace Material		Maximum Amount from
<u>Name</u>	Symbol	Mercury Tests, ppm
Freon 114	F114	6,000
Ethylene Dichloride	EDC	40
Vinylidene Chloride	VDC	2
Methylene Chloride	MC	2
Benzene	BEN	1
Ethyl Alcohol	EAL	3

Trace materials common to both atmospheres are present at roughly a three-fold higher concentration in the Mercury capsule atmosphere than in the Apollo C/M atmosphere on the basis of the SCS contribution to trace material generation. Therefore, it is estimated that there will be approximately three times more electronic gear than represented by the SCS contributing to trace material contamination. Accordingly, the generation rate profile of Table 7 can be multiplied by three and the new trace materials may be added to the list in the proportion found in Mercury capsule atmospheres. This provides the complete rate generation pattern for materials, processes, and equipment.

Specifically, the amount of ethyl alcohol from Table 7 is 203 micromoles. If the total amount of alcohol is three times this value, the amounts of new materials are as reported in Table 9.

AMOUNTS OF TRACE CONTAMINANTS GENERATED BY OTHER THAN
ELECTRONIC GEAR (14-DAY MISSION)

TABLE 9

Trace <u>Contaminant</u>	Relative Amount	Actual Amount, Micromoles
F114	6,000	1,218,000
EDC	40	8,120
VDC	2	406
MC	2	406
BEN	1	203
EAL	2	609

Volatile Materials of Biological Origin

In assessing the identity and amount of volatile materials to be expected from biological sources, considerable reliance has been placed on secondary sources of information such as texts, the <u>Bioastronautics Data Book</u>, and the <u>Biology Data Book</u>. Efforts have, however, been made to search out significant additions to the literature which have appeared since 1960, and to return to the original literature to evaluate particularly important conclusions.

Sources of atmospheric contamination considered, and disposition made of the sources are tabulated in Table 10.

TABLE 10

DISPOSITION OF BIOLOGICAL SOURCES OF ATMOSPHERIC CONTAMINATION

Source

Disposition

Feces

Not considered a significant source of atmospheric contamination since feces are to be deposited in a gas-proof bag which is immediately sealed.

Urine

Same as for feces.

Skin Secretions

It is assumed that all skin secretions, including sweat and sebum, reach the space capsule atmosphere except as limited by their volatility.

Intestinal Gases

Assumed that all flatus gases reach the atmosphere of the space capsule.

Breath

Assumed that all respiratory gases reach the space capsule atmosphere.

Food

Since Food is to be packaged in hermetically sealed containers to be opened only as the food is consumed, it is assumed not to make significant contributions to the atmospheric contamination. Experience has shown that food odors can be noticeable with extremely small actual concentrations of odor components. The marked odors of many food items might suggest that significant contamination of the atmosphere with compounds from the food would occur even in the brief exposure to air that occurs when the packages of food are opened for consumption. Attempts to identify the flavor components of food, however, have served to emphasize the exquisite sensitivity of the olfactory apparatus. detectors, such as are used in gas chromatography, can detect parts per billion, but the olfactory nerves can detect 100 to 10,000 times less(30,31). The number of compounds actually involved in food odors have proved extremely numerous (32,33).

Hygiene Activities

Since these are to be limited to provisions for towelling, they are assumed to make no significant contributions.

Miscellaneous Sources

Tears
Nasal Drippings
Hair
Nail Clippings
Ear Wax

None considered significant sources of atmospheric contamination.

In estimating atmospheric contributions, an effort has been made to keep the estimates within what seem reasonable ranges, but to tend toward high rather than low estimates. This is, of course, in the interest of safety. All rate estimates are on a per man basis.

Information is almost totally lacking in some areas, and this is pointed out in the discussion. In some cases, it is possible that a very extensive effort in searching the literature would supply additional useful data.

Skin Secretions

Skin secretions include eccrine sweat, apocrine sweat, and sebum.

Added to these are materials contributed by sluffing of the surface layer of cells, and materials added by gas exchange through the skin. Finally, bacterial action may play an important part in altering the identity of the materials after they appear at the skin surface. The magnitude of these alterations will be dependent on body hygiene and the means taken to cope with actual runoff of sweat.

It is assumed that only volatility will limit the contribution of skin secretions to the atmosphere. This is, of course, an extreme assumption, conservative on the side of safety. If skin secretions are physically removed and disposed of, they will to that extent not make contributions to the atmosphere. Skin secretions which are absorbed on clothing may or may not continue to make atmospheric contributions, depending on the disposition that

is made of the clothing. The wearing of light clothing may not markedly reduce the amount of volatile skin secretions escaping into the atmosphere.

As the clothing becomes saturated, it presents a large surface for evaporation.

Eccrine Sweat. Eccrine sweat is that part of the total skin secretions contributed by thermally responsive sweat glands. These are unevenly distributed over the body surface, there being about six times the density of sweat glands on the palms of the hands as on the back and buttocks. The composition of eccrine sweat is not constant, but is, among other influences, a function of the sweating rate. Rothman⁽³⁾, for example, notes nitrogen concentrations of 68 mg per 100 cc in profusely emitted sweat and 275 mg per 100 cc in intermittently secreted sweat. These effects are at least in the direction of rendering total secretion of solutes less dependent of sweat volume.

Measurements of sweat composition have frequently been made by surrounding a body member with an impermeable bag and collecting all aqueous secretions for analysis. Other collection techniques have sometimes been used. Accordingly, measurements of sweating rate are subject to wide variation, and values from near zero to over three liters per hour have been noted. The recent data of Consolazio and co-workers⁽⁴⁾ seem a good basis on which to make estimates for present purposes. He conducted studies at a series of temperatures and for an activity profile in which most of the day was spent

in sedentary activity except for two exercise periods of 50 minutes each.

Sweating rates were:

Mean Temp.,	Sweat Rate
70	143
85	242
100	312

Using these values together with values cited in the <u>Bioastronautics</u>

<u>Data Book</u> (5), the following sweating rates have been selected as a basis

for calculation of atmospheric contamination from sweat:

Low	50	g/hr
Median	250	g/hr
High	400	g/hr

The median value is used in calculating the amounts of individual trace materials discussed below.

Lactic Acid. Lactic acid appears in sweat in relatively large amounts and has significant volatility. It must, therefore, be considered a potential contaminant. The amount in sweat has been shown by Astrand (6) to go as high as 500 mg per 100 ml during the first of a sweating period, but to level off at about 156 to 190 mg per 100 ml as sweating continues. Using 190 mg per 100 ml for purposes of calculation and using sweating rates calculated above, the total lactate secretion might be:

Low	95	mg/hr
Median	475	mg/hr
High	760	mg/hr

The total amount of this supply which will evaporate into the atmosphere is limited by the amount present as lactic acid itself, by the concentration reached as the water component of sweat evaporates, by the temperature, and by the surface area from which evaporation occurs.

The amount present as lactic acid is a function of pH. Skin pH values do not ordinarily vary by more than one pH unit and pH 5.7 represents a good estimate of the mean. At pH 5.7 most of the lactic acid will be present as the lactate ion, since the ionization constant of lactic acid is 1.5×10^{-4} (pKa = 3.81). Only about one percent of the lactic acid will be present in nonionized form at pH 5.7, and only about 10 percent at pH 4.7. The vapor pressure of pure lactic acid at 40° C would be about 0.25 mm. This means that at equilibrium the air above lactic acid at 40° will contain 1.4 x 10^{-4} mole per liter.

If nearly all the aqueous phase of sweat evaporates it could leave behind a system containing lactic acid and lactates. Limiting concentrations of lactic acid will then be:

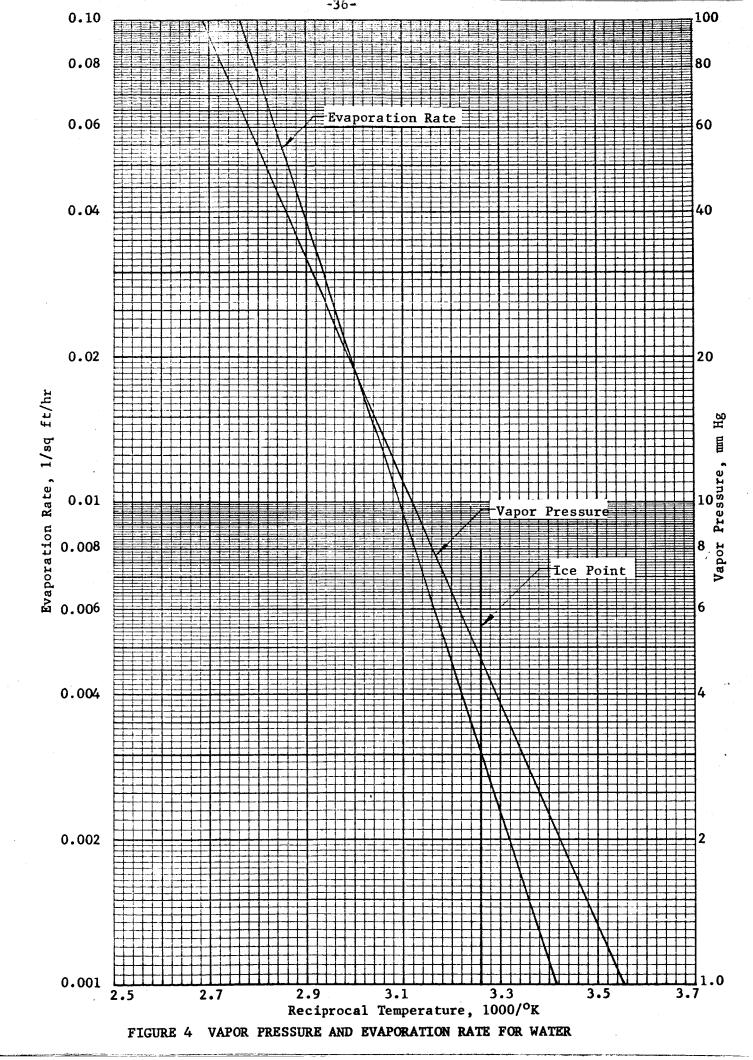
at pH 4.7 - 14 µ mole/1 gas at pH 5.7 - 1.4µ mole/1 gas

If new sweat accumulates so that the limiting concentration on the evaporative surface is 1 m, then the mole fraction of total lactate will be about 0.02 and the limiting concentrations will be:

at pH 4.7 - 0.28µ mole/1 gas at pH 5.7 - 0.028µ mole/1 gas The latter conditions seem the likely ones, and a limiting concentration of 0.05 μ mole per liter of atmosphere seems a safe estimate.

It is instructive to estimate the probable rate of evaporation of free lactic acid should it by chance be present in greater amounts than indicated above. This was done by noting the constant proportionality between measured rates of evaporation and vapor pressures for the case of water (Figure 4) down to the ice point. If evaporation rate is proportional to vapor pressure, and if the latter declines exponentially with decreasing temperature, it is apparent that the evaporation rate of slightly volatile acids will be very low at body temperature. For the case of lactic acid having a vapor pressure of two mm Hg at 57°C, the evaporation rate would be about one g per square foot per hour. This is the rate of evaporation of water having a vapor pressure of two mm Hg. At body temperature, the vapor pressure of lactic acid would be near 0.25 mm Hg. The corresponding evaporation rate would be near 0.06 g per square foot per hour or 1.2 g per man-hour as a maximum. In consideration of the fact that less than one percent of the acid is free acid, and that vapor pressure would be lowered by a factor of 50 for 0.02 mole fraction solution, this maximum rate would be reduced by a factor of about 4000 to 2.4 x 10⁻⁴ g per man-hour or 2.5 µ moles per man-hour.

This is a borderline rate for atmospheric contamination. In view of the probability that there will be even less than one percent free lactic acid, this acid will not be given further consideration as a trace contaminant.



Ammonia. "Ammonia" is present in only minimal quantities in freshly excreted sweat⁽¹⁾. It is formed by bacterial decomposition of urea. The amount found on examination of sweat usually ranges from 2.5 to 35 mg per 100 ml⁽⁵⁾. If 20 mg per 100 ml is used for purposes of calculation, then estimates of amounts produced would be:

Low 10 mg/hr Median 50 mg/hr High 80 mg/hr

The actual concentration of ammonia as such will be a function of the pH of the system. The ionization constant of ammonium hydroxide is 2.5 x 10⁻⁵ at 40°C. The pH at the equivalence point will, therefore, be about 9.25. This means that at the pH prevailing on the skin (5.7) about 0.06 percent of the total ammonia of the system will be nonprotonated. The volatility will be limited by concentration and by pH shifts which occur as ammonia is lost from the system.

If evaporation of water occurs so rapidly that the skin remains essentially dry, then the contribution to the atmosphere might be estimated at less than the following values:

Low 0.0006 x 10 mg/hr = 0.006 mg/hr = 0.35 μ mole/hr Median 0.0006 x 50 mg/hr = 0.03 mg/hr = 1.7 μ mole/hr High 0.0006 x 80 mg/hr = 0.048 mg/hr = 2.8 μ mole/hr

Since the skin will nearly always be fairly moist, actual rates will be 0.1 to 0.0002 of these values. Rates of 0.02µ mole per hour might represent a reasonable estimate.

These calculations assume that the anions are nonvolatile. This may not be entirely correct. Dissociation of weak salts to yield volatile acids and ammonia could raise these values somewhat.

Since ammonia at the body surface is largely contributed by urea decomposition, estimates of the potential supply are of some pertinence. It is possible that bacterial action on sweat-soaked clothes where the buffering action of new sweat deposits was not available might contribute substantial quantities of ammonia. The amount of urea in sweat expressed in terms of urea nitrogen is given as 5 to 39 mg per 100 ml by <u>Bioastronautics</u> <u>Data Book</u> (5). A paper by Brusilow and Gordes (7) gives data covering the range 13.5 to 40 mg per 100 ml.

If 30 mg of potential ammonia per 100 ml sweat is used for purposes of calculation, estimates of rates become:

Low 15 mg/hr = 0.89 m mole/hr Median 75 mg/hr = 4.2 m mole/hr High 120 mg/hr = 7.0 m mole/hr

Phenol. The amount of phenol in sweat is given in the <u>Bioastro-nautics Data Book</u> (5) as two to eight mg per 100 ml of sweat. The source of these data is apparently Altman and Dittmer (8). Altman and Dittmer

reference their source incorrectly, but do list in their references a source which refers in turn to a paper by Deichmann and Schafer (9). These workers found 0.09 to 0.44 mg of free phenol and 0.04 to 0.11 mg of conjugated phenol per 100 ml of sweat. Total concentrations were from 0.16 to 0.55 mg per 100 ml. These values are an order of magnitude lower than those cited by Altman and Dittmer.

Using 0.4 mg per 100 ml for making estimates of phenol from sweat one obtains the following figures:

Rate = 0.4 mg/100 ml

Low 0.2 mg/hr = 2.1µ mole/hr Median 1.0 mg/hr = 10µ mole/hr High 1.6 mg/hr = 17µ mole/hr

Phenol has a vapor pressure of one mm Hg at 40°C. Moreover, the water-phenol system shows positive deviations from ideality, thus leading to effective removal of phenol by aeration, or distillation. It is, therefore, expected that phenol will be evaporated along with sweat. The figure of 10µ mole per man-hour is used for calculation of contamination rates.

Iodine. Iodide ion is a potentially volatile material since it may be gradually oxidized to iodine by exposure to atmospheric oxygen. The total amount available is, however, judged too small to be of importance. The concentration in mg per 100 mg sweat is given as $0.0009^{(5)}$. At the median sweating rate of 250 ml per hour, this would give only 2 mg per three-man mission of 14 days.

Apocrine Sweat. Apocrine sweat is produced by glands located in the axillae, perimamillary regions, mid-line of the abdomen, mons pubis, perigenital, and perineal areas and in the external ear, and the nasal vestibulae. These glands do not respond to thermal stimulation but do respond to mental stimuli via the adrenergic system⁽³⁾. Data on the composition of apocrine sweat are minimal. Rothman⁽³⁾ states that free fatty acids are present even in the absence of bacterial action. Limited data gathered by Thurmon and Ottenstein⁽¹⁰⁾ indicate pH values about one-half unit higher than eccrine sweat, and lactate concentrations about equivalent to eccrine sweat. Ottenstein⁽¹¹⁾ gives figures indicating ammonia levels several times higher than for eccrine sweat but with lower urea values. One suspects bacterial action.

Kligman and Shebackh⁽¹²⁾ report that apocrine sweat becomes odorous only as a result of bacterial action. They also report the apocrine glands of pubic regions to be nonfunctional.

From the limited data available one might expect that the apocrine sweat would contribute principally to additional ammonia, probably to the extent of 0.05µ mole per hour. This is negligible compared to the amount produced by bacterial action.

Insensible Sweat. Insensible sweat arises by unnoted evaporation of eccrine and apocrine sweat, and by water exchange directly through the skin. Methodology used to measure total sweat would include the volatile components of insensible sweat within the resulting estimates.

Sebum. Sebum consists of fat and cellular debris on the skin surface. A main source is the sebaceous glands associated with hair follicles, present in man everywhere hair follicles are present. Other sources are apocrine glands and the keratinizing epidermis. Analytical examination has usually been confined to the lipid fraction. The amount of sebum produced is limited by the amount already present on the skin and by the temperature. Sebum tends to solidify in the gland openings and prevent further secretion until it is removed by physical means or by emulsification in sweat (13). One is, therefore, concerned both with saturation levels and with rates of production. As in the case of sweat, material absorbed on clothing continues to be available for evaporation. The melting point of sebum is in the vicinity of 30°c (13).

Saturation levels of sebum, where precautions to prevent removal are taken, have been reported at from 0.38 mg per sq cm to 3.38 mg per sq cm, with more of the values at the upper part of the range (3). Levels observed without precautions to prevent removal have varied from 0.14 to 0.24 mg per sq cm (14,15). These values are all based on measurements on the forehead. Values for other parts of the body may be only one-fifth as great. Some data cited by Carruthers (14) are:

Forehead

Vertebrae line
Nudiapsular lines

Vertebrae line

Vertebrae line

Nudiapsular line

0.75 mg/cm

Nudiapsular line

0.50 mg/cm

Lower back

Some production rates available in the literature are:

1.69
$$\pm$$
 0.52 mg/10 cm²/hr = 0.054 mg/cm²/hr, (15)
0.3 to 1.8 mg/20 cm²/hr = 0.015 to 0.09 mg/cm²/hr, (13)
and 0.05 mg/cm²/hr. (14)

Minimum rate

0.1
$$\%/\text{cm}^2/\text{min} = 0.006 \text{ mg/cm}^2/\text{hr}^{(3)}$$

From these various considerations the following are estimated for purposes of calculation:

Casual levels

Low	0.15	mg/cm ² mg/cm mg/cm
Median	0.3	mg/cm2
High	0.9	mg/cm ²

Production rates

Low	0.01	mg/cm ² /hr mg/cm ² /hr mg/cm ² /hr
Median	0.02	mg/cm2/hr
High	0.05	mg/cm ² /hr

Lower Fatty Acids in Sebum. Butyric, valeric, and caproic acids are reported to be present in skin secretions to the extent of 1.2 percent. (5)

This estimate seems extremely doubtful. The source from which it comes (16) quotes the chemical analysis of sebum as:

Water	31.7 percent
Epithelium and protein	61.75 percent
Fat	4.16 percent
Butyric, valeric, and caproic acids	1.21 percent
Ash	1.18 percent
Butyric, valeric, and caproic acids	1.21 percen

The methods of sebum examination usually used $^{(17,18)}$ begin with organic-solvent extractions to remove the lipid, followed by examination of the materials so obtained. If butyric, valeric, and caproic acids were really present in large amounts, one would expect them to be extremely prominent in such extracts. Haahti, $^{(17)}$ using careful gas chromatographic techniques, noted only traces of acids below C_{12} . Boughton and Wheatly, $^{(18)}$ using gas chromatographic methods, placed the amount of fatty acids of C_{10} and lower at 0.2 percent. If these acids are really present in large amounts, one would expect their salts to be leached into eccrine sweat, but they have not appeared prominently in analysis of this material.

The data giving butyric, valeric, and caproic acids as 1.2 percent of total skin secretions comes from Sunderman $^{(16)}$ as a tertiary source. He identifies the original source as a 1912 text. $^{(19)}$

Only traces of fatty acids C_7 to C_9 are present in sebum; the amounts are individually less than 0.05 percent. (14)

Data quoted by Carruthers $^{(14)}$ give the amount of n-decanoic acid as 0.08 percent of the total free fatty acids of sebum in one case, and the total amount of free C_{10} acids as 0.3 percent of the total free fatty acids of sebum in another case. About 30 percent of the total weight of sebum is free fatty acids. $^{(14,20)}$

The total amount in sebum using assumptions previously quoted in the general discussion of sebum will be:

Low 0.15 mg sebum/cm² x 19,000 cm² x 0.30 x 0.003 = 2.5 mg Median 5.0 mg High 15 mg

The production rate is estimated as follows:

Low $0.01 \text{ mg/cm}^2/\text{hr} \times 19,000 \text{ cm}^2 \times 0.30 \times 0.003 = 0.17 \text{ mg/hr}$ Median 0.34 mg/hrHigh 1.0 mg/hr

The actual amount reaching the atmosphere will be a function of its intrinsic volatility and its dilution by other components of the system. The vapor pressure of n-decanoic acid at 125°C is one mm Hg and at 268°C is 760 mm Hg. Extrapolation of the vapor pressure curve predicts a vapor pressure of only 0.01 mm at 37°C indicating a negligible evaporation rate. At one atmosphere pressure the limiting concentration is therefore one part by volume to 76,000 parts air or 80 ppm by weight. Since the acid makes up only about 0.09 percent of sebum and since the molecular weight of other components probably averages not more than three times that of decanoic acid, the limiting concentration can be estimated at

80 ppm x 0.0009 x 3 = 0.21 ppm.

This will be further reduced by the fact that the pH of skin will reduce the amount of nonionized fatty acid to about 15 percent of the total (pKa = 4.9).

It seems obvious that the contribution of decanoic acid will be negligible.

Higher Fatty Acids in Sebum. The percentage of total free fatty acids of sebum represented by the prominent higher members of the series is given by sources quoted in Carruthers (15) as follows:

n-C ₁₂	3.6	percent
n-C ₁₄	6.3	percent
n-C ₁₆	24.2	percent
Oleic	35.6	percent

Data on the volatility of these acids follow:

	Temperature, ^O C						
Acid	<u>1 mm</u>	10 mm	40 mm	100 mm	400 mm	760 mm	
n-C ₁₂	121	166	201	227	273	299	
n-C ₁₄	142	199	223	250	294	318	
n-C ₁₆	153	205	244	271	326	353	
Oleic	176	223	257	286	334	360	

Projecting these figures graphically, the vapor pressure of n-C₁₂ at 37°C is predicted at about 0.002 mm, and that of oleic acid at 0.00001 mm. Therefore, rates of evaporation would be negligible. The limiting concentration of pure lauric acid would be 0.000021 g per liter. At one atmosphere, this is 16 ppm. When the effects of concentration and of pH are taken into consideration, the equilibrium concentration is certainly not greater than 300 ppb. The equilibrium concentrations of the other acids considered in this group is still less.

Potential Fatty Acids in Sebum. The potential supply of fatty acids formed by hydrolysis of the glycerides of sebum would about double the free fatty acid fraction. Hydrolysis would free the acids of their chemical association with glycerin. However, the hydrolysis products probably would be buffered by ammonia and would not remain as free undissociated acid.

Flatus

Flatus will represent a major source of gaseous contamination of the space vehicle. The amount of flatus depends markedly upon the dietary composition, but those foods which promote flatus production are known, and it is assumed that they will be avoided in assembling the astronaut's diet. The composition of flatus varies markedly between individuals. It has been found that some people regularly produce hydrogen and others do not. Similarly, some persons produce methane and others do not. Murphy (21) states that about half the subjects he has studied produce methane and half do not.

Flatus egestion is not constant, but periodic. Murphy (21) estimates one 20-50 ml egestion per hour.

Some figures on flatus volume and composition are given in Table 11A.

TABLE 11

VOLUME AND COMPOSITION OF FLATUS

A. Volume Per Man

	m1/1	Day	Base for			
Author	<u> Average</u>	Range	Calculation	Reference		
Kirk	2140	520-5720	ml/min 12-hr day ^(a)	22		
Askevald	980 250-600			23		
			m1/min	23		
Beazell, Ivy	527	3 80- 655		24		
Murphy		480-1200	"20-50 m1/hr"	21		
		100-500		21		

B. Methane in Flatus

Author	<u>Concentra</u> <u>Average</u>	Range	Estimated Rate of Production	Reference
Kirk	7.2	0-30		22
Askevald	0.4 ± 0.1			23
Callaway	_		0-23 m1/hr	25
Murphy	18	0-40		21
Murphy			0-173 ml/day	26

C. Hydrogen in Flatus

Author	Concentra Average	Range	Estimated Rate of Production	Reference
Kirk	20.9	3-34		22
Askevald	2.3 ± 0.3	5		23
Callaway	_		0-16 m1/hr	25
Murphy	9	0-50		21
Murphy			0-24 ml/day	26

⁽a) Quoted range evidently for 12-hour day.

On the basis of these figures the following volume rates of production are used as a base for calculation:

> Low 150 ml/day = 6.2 ml/hr Median 600 ml/day = 25 ml/hr High 1500 ml/day = 62 ml/hr

Methane. The amount of methane produced varies widely with individuals ranging from essentially none to several ml per hour. Some basis for estimates are given in Table 11-B.

Using Murphy's figure of 18 percent methane and applying it to volume estimates quoted earlier gives the following rates of production of methane:

Low 27 ml/day = 1.1 ml/hr Median 108 ml/day = 4.5 ml/hr High 270 ml/day = 11 ml/hr

These seem reasonable estimates in view of the data at hand.

It is to be remembered that the total methane addition to the atmosphere each hour may come in increments comprising the total hour's production, or even the production of several hours.

<u>Hydrogen</u>. The amount of hydrogen produced varies widely with individuals and may be none. Some basis for estimates are given in Table 11-C.

Using Murphy's figure of 9 percent average hydrogen in flatus and applying it to the volume estimates quoted earlier gives the following volume rates of production of hydrogen:

Low 13 ml/day = 0.54 ml/hr Median 54 ml/day = 2.2 ml/hr High 130 ml/day = 5.4 ml/hr

These seem reasonable estimates in view of the data available.

Hydrogen Sulfide. The amount of hydrogen sulfide in flatus has been inadequately determined. Kirk $^{(22)}$ gives an average value of 0.00028 percent on a cabbage-free diet. Murphy $^{(26)}$ has estimated the amount at certainly less than 0.0025 percent. Using the higher figure for safety estimates of generation rate are:

Low 0.0037 ml/day = 0.00015 ml/hr Median 0.0150 ml/day = 0.00062 ml/hr High 0.0370 ml/day = 0.0015 ml/hr

<u>Unidentified Constituents</u>. Murphy (26) has estimated unidentified constituents of flatus as being present in concentrations possibly as high as one percent.

Respiratory Gases

Hydrogen and methane occur in respiratory gases but as reflections of hydrogen and methane of flatus. (26) It is, therefore, probably not necessary that separate allowances be made for these gases from respiratory sources.

Because of the large volume of respiratory gases, the presence of even extremely small concentrations of materials in these gases could make very important contributions to air contamination in a closed space. The elimination of acetone in diabetes, of alcohol following ingestion, of ether and other organic solvents following administration, and of selenium-containing compounds by the respiratory route is well known. More attention to respiratory gases as a source of contamination would seem well warranted.

Carbon Monoxide. Carbon monoxide has been identified in expired air and hemoglobin breakdown has been incriminated as a source. (27-9) In three subjects the volume produced was estimated at 0.5, 0.65, and 1.05 ml per hour. (29) Concentrations in the expired air were from 0.00013 to 0.00017 percent. In a study of hospitalized subjects concentrations were 0.00023 \pm 0.00001 percent. (28)

Estimates for production per day might be:

Low 10 ml/day = 0.41 ml/hr Mediam 20 ml/day = 0.84 ml/hr High 30 ml/day = 1.2 ml/hr

Illness or infection might increase the amount.

Summary

Estimates of contributions to atmospheric contamination by materials considered in this literature search are summarized in Table 12. Converted to generation rates based on three men, these data represent the addition of new or additional trace contaminants in the amounts given in Table 13.

TABLE 12

PRODUCTION PER MAN OF ATMOSPHERIC CONTAMINANTS

Limiting Conc.**, µ Moles/1	0.05					•	0.03	0.2					
les/Hour High			7000	17		1e			490	240	0.068	28	23
Rate of Production*, u Moles/Hour Low Median	Negligible	0.02	4200	10	Negligible	Unknown, probably negligible	Negligible	Negligible	200	100	0.028	11	38
Rate of Low			890	7		Unknown,	•	7	65	24	0.007	m	18
Source	Eccrine sweat	Eccrine sweat	Urea decomposition by bacteria	Eccrine sweat	Eccrine sweat toxication	Sebum	Sebum	Sebum	Flatus & breath	Flatus & breath	Flatus	Flatus	Carbon monoxide
Substance	Lactic Acid	Ammonta	Ammonia	Pheno1	Iodine	Butyric, Caproic, Valeric Acids	Decanoic Acid	Lauric Acid	Methane	Hydrogen	Hydrogen Sulfide	Unidentified Gases	Carbon Monoxide

^{*} Numbers refer to amount estimated to reach the atmosphere.

^{**}Based upon attainment of equilibrium vapor pressure.

TABLE 13

TRACE CONTAMINANTS FROM BIOLOGICAL SOURCES DURING 14-DAY MISSION

Trace Contaminant	Symbol	Amount <u>Micromoles</u>
Ammonia	NH ₃	4,200,000
Me thane	СН ₄	206,000
Hydrogen	н ₂	100,800
Carbon Monoxide	со	53,900
Pheno1	PH	10,080

Selecting Trace Materials for Use in Trace Material Control Unit

A numerical comparison of toxicity ratings for trace materials was required in order that substitutions could be made for the 12 trace materials from the Stabilization and Control System to arrive at a simulation program based on a lesser number of trace materials. There is no established procedure for combining all toxicity data related to a given chemical species to arrive at a relative measure of toxicity. Yet, some scheme had to be devised to obtain an over-all rating for each compound so that trace contaminants could be given a numerical classification. The classification developed for this purpose was based upon the following factors:

TLV Rating The threshold limiting value set by the American Conference of Governmental

Industrial Hygienists.

Toxicity Rating* 0 - No harm.

1 - Slight reversible changes.

2 - Moderate reversible or irreversible changes not causing permanent injury.

3 - Severe effects possibly causing death or permanent injury after short exposure to small quantities.

Site Rating Rated as 1 if local.

Rated as 2 if systemic.

Speed Rating Rated as 1 if chronic.

Rated as 3 if acute.

Route Rating Rated as 1 if by inhalation or if an irritant.

Rated as 5 if by absorption. Rated as 10 if by ingestion.

^{*}As rated by Sax, N. Irving, "Dangerous Properties of Industrial Materials", Reinhold, 2nd Edition (1963).

The ratings for site, speed, and route are original to this program, as is the summation of ratings given below over multiple modes of action:

"Relative Toxicity" =
$$\frac{10,000}{\text{TLV}} \sum \frac{\text{(Toxicity) (Site) (Speed)}}{\text{Route}}$$

An example of the use of the above equation is given for the case of carbon monoxide which has a TLV of 100, and the following toxicity ratings:

Toxicity Ratings for Carbon Monoxide

		Acute		Ch ₁	ronic
	Local		Systemic	Local	Systemic
(a)	(b)	(c)	(d)	(e)	(f)
Irritant	Ingestion	Inhalation	Inhalation	Irritant	Inhalation
0	0	0	3	0	1

Summing for carbon monoxide:

$$\begin{array}{c} \text{(Toxicity)} \\ \text{(Rating)} \end{array} \times \\ \begin{array}{c} \text{(Site)} \\ \text{(Rating)} \end{array} \times \\ \begin{array}{c} \text{(Route)} \\ \text{(Rating)} \end{array}$$

(a) No harm (0) x local (1) x acute (3)
$$\frac{.}{.}$$
 irritant (1) = $\frac{0 \times 1 \times 3}{1} = 0$

(b) " "
$$x$$
 " $\frac{\cdot}{\cdot}$ ingestion (10) = $\frac{0 \times 1 \times 3}{10} = 0$

(c) " " x "
$$\frac{\cdot}{\cdot}$$
 inhalation (1) = $\frac{0 \times 1 \times 3}{1} = 0$

(d) Severe (3) x systemic (2) x "
$$\frac{1}{1}$$
 inhalation (1) = $\frac{3 \times 2 \times 3}{1}$ = 18

(e) No harm (0) x local (1) x chronic(1)
$$\frac{1}{1}$$
 irritant (1) = $\frac{0 \times 1 \times 1}{1} = 0$

(f) Slight (1) x systemic (2) x "
$$\frac{1 \times 2 \times 1}{1} = 2$$
 $= 20$

Thus, for carbon monoxide,

"Relative Toxicity" =
$$\frac{10,000}{\text{TLV}} \ge \frac{10,000}{100} \times 20 = 2,000$$
.

Table 14 is a compilation of toxicity criteria for trace materials. The data were used to obtain "relative toxicity" ratings for other materials in the manner illustrated above for carbon monoxide. The normalized ratings are given in Table 15. These ratings were used to make substitutions for some of the trace materials to arrive at a simulation program for the entire inventory based on a lesser number of trace materials. Substitutions were made within chemical families where possible. This was done in deference to the opinion that numerical toxicity ratings are not truly quantitative and should be used with due regard taken of the known chemical and physiological properties of each material.

CHARACTERIZATIO

Compound (List includes those appearing once or more as major constitu- ents & 5 or more times as minor constituents)	<u>Formula</u>	Prevalence Ranking*	Molecular Weight	Tempera- ture 760 mm (°C)	Dipole Moment (µx 10 ¹⁸ esu)	TLV (PPM)	<u>Acı</u> — <u>I</u> 1
Carbon Monoxide	co	1	28	-192	0.1	100	
Carbon Dioxide	CO ₂	2	44	-78s	0	5000	
Acetone	CH3COCH3	. 3	58.1	56	2.89	1000	
Ethyl Formate	HCOOC ₂ H ₅	3	74.1	58	1.93	100	
Methyl Acetate	СН ₃ СО <mark>2</mark> СН3	3	74.1	54	1.72	200	
n-Butyl Alcohol	CH3 (CH2) 2CH2OH	4	74.1	117	1.67	100	
Methyl Isobutyl Ketone	(CH3)2CHCH2COCH3	5	100.2	1 18	(2.8)	100	
m-Xylene	C ₆ H ₄ (CH ₃) ₂	6	106.2	139	(0.3)	200	
o-Xylene	C6H4 (CH3)2	6	106.2	14 4	0.62	200	
p-Xylene	C6H4 (CH3)2	6	106.2	138	0	200	
Ethyl Alcohol	CH3CH2OH	7	46.1	79	1.70	1000	
iso Propyl Alcohol	СН3СНОНСН3	8	60.1	82	1.60	400	
Toluene	C6H5CH3	9	92.1	111	0.36	200	
Propionaldehyde	СН3СН2СНО	10	58.1	49(740)	2.72	?	
Benzene	C6H6	11	78.1	80`	0	25	
Ethylben ze ne	C6H5C2H5	12	106.2	136	0.59	200	
Trichloroethylene	CHCLCCL2	13	131.4	87	(1.5)	100	
Carbon Disulfide	CS ₂	14	76.1	46	ò	20	
Acetaldehyde	CH ₃ CHO	15	44	21	2.72	200	
n-Valeraldehyde	СДНоСНО	16	86.1	103	(2.72)	?	
n-Butyraldehyde	CH3(CH2)2CHO	17	72.1	76	2.72	?	
Methylethyl Ketone	CH3COCH2CH3	18	72.1	80	(3)	200	
tert-Butyl Alcohol	(CH3)3COH	19	74.1	83	16	100	
Heptane	СH3(CH2)5CH3	20	100.2	98	0	500	
iso Butyl Alcohol	(CH ₃) ₂ CHCH ₂ OH	21	74.1	108	1.64	?	
Mesitylene	C9H12	22	120.2	165	0	?	
Methyl Methacrylate	CH ₂ C(CH ₃)COOCH ₃		100.1	101.0	(1.8)	400	
Freon 114	CCLF2CCLF2		171	3.5	0	1000	
Ethylene Dichloride	CH2CLCH2CL		99.0	83.5	Ö	50	
Vinylidene Chloride	CH2CCL2		97.0	31.6	(2.05)	500	
Methylene Chloride	CH ₂ CL ₂		84.9	40.1	0	500	
Ammonia	NH3		17.0	-33.35	1.46	100	
Methane	CH ₄		16.0	-161.5	0		
Hydrogen	H ₂		2.0	-259	ŏ		
Phenol	n ₂ С ₆ н ₅ 0н		94.1	181.9	1.7	5	

^{*} Prevalence Ranking -- those which are rated appeared once or more as major constituents and 5 materials at 200°F in 5 PSIA oxygen. Reference 2.

^{**}Data from Sax, "Dangerous Properties of Industrial Materials", Reinhold, 2nd Edition (1963).

^{0 -} No harm.

^{1 -} Slight-readily reversible.2 - Moderate-reversible or irreversible. No death or permanent injury.

LE 14

OF TRACE MATERIALS

Toxicity Criteria** Chronic (long duration, te (short duration exposure or dose-seconds, minutes, days, months, years) Local Systemic Loca1 Systemic Skin Skin ritant Ingestion Inhalation Ingestion Inhalation Absorption Irritant Ingestion Inhalation Absorption Unknown U U U U ĺ U U U U U U U U U U U U U 2 2 3 U U U U U U U U U U U U Ħ Ħ U U U U U U П U O

or more times as minor constituents among compounds outgassed from certain organic engineering



^{3 -} High -- may cause death or permanent injury with small dose.

U - Unknown -- no information on humans considered valid.

TABLE 15
"RELATIVE TOXICITY" RATINGS OF TRACE CONTAMINANTS

		Ilm et
Trace Contaminant	Symbol	"Relative <u>Toxicity"*</u>
Carbon Monoxide	СО	100
Acetone	AC	14.4
Ethyl Formate	AC +	120
Methyl Acetate	vc_{+}	60
n-Butyl Alcohol	$NB\Lambda L$	110
Methylene Chloride	MC	37
Methyl-n-Butyl Ketone	MIBK	160
o- m- or p-Xylene	XYL	30
Ethyl Alcohol	EAL	10.5
i-Propyl Alcohol	IPAL	28
Toluene	TOL	62
Benzene	Ben	759
Ethyl Benzene	EBE	62
Trichloroethylene	TCE	143
Carbon Disulfide	CDS	952
Acetaldehyde	ACA	95
Methyl Ethyl Ketone	MEK	48
t-Butyl Alcohol	TBAL	95
Methyl Methacrylate	MMC	9.4
Cumene	CUM	30 (est.)
Freon 114	F114	7.5
Ethylene Dichloride	EDC	480
Vinylidene Chloride	VD C	23 (est.)
Ammonia	nн ₃	36.8
Methylene Chloride	мс	37 (based on vinyl chloride)
Me thane	CH ₄	**
Hydrogen	н ₂	\\\\
Pheno1	PH	2000

^{*} Normalized to a value of 100 for carbon monoxide.

^{**}Not rated -- simple asphoxiants offering fire and explosion hazards.

⁺ Materials expressed also as acetone.

Generation Rate Patterns

The generation rate pattern for the SCS will be used to illustrate the method of simulating generation rate patterns with the trace material control unit. The amounts of trace materials and their "relative toxicity" ratings are compiled in Table 16 from corresponding data in Tables 8 and 15. Substitute trace materials and their "relative toxicities" are also listed in Table 16, and a final summation is made over the quantities of substitute materials. This procedure limited to five the number of trace materials required to program the control unit for simulation of generation rates from the SCS.

TABLE 16.

FINAL GENERATION RATE PATTERN FOR SCS SYSTEM (14-DAY TOTALS)

عا	Actual Trace Materia Amount, "R	erial "Relative	Substi Name or	Substitute Trace Material e or "Relative Amoun	faterial Amount,	Sum Substitute	Summation Over Substitute Trace Materials
Micromoles	oles	Toxicity"	Symbol	Toxicity"	Micromoles	Mater 1a1	Amount, µ Moles
7,510	10	28	NBAL	110	1,910	Ace tone	38,000
5,4	5,465	100	**00	100	5,465	8	5,465
Ε	3,109	14.4-120	Acetone	14.4	26,000	NBAL	2,064
1,	1,044	160	Acetone	14.4	11,600	HEP	480
	720	30	BEN***	759	29	BEN	86
	672	63	BEN	759	99	UNK	(1,180)
	480	19	HEP**	19	480		
	203	10.5	NBAL	110	20		
	134	110	NBAL**	110	134		
	105	30 est.	BEN	759	4		
	102	62	BEN	759	6		
	90	87	Acetone	14.4	310		
	40	5.6	Acetone	14.4	61	_	
ų	1,180	8 1	;	:	(1,180)		

* See Table 14 for names corresponding to these symbols.

^{**} Actual trace material used in TMCU.

^{***}Benzene.

It was shown in the section of this report on "Trace Materials Not Generated by SCS" that there will be approximately three times more electronic gear than represented by the SCS contributing to trace material contamination. It is apparent that a complete generation rate pattern will result from multiplying the generation rate profile for the SCS by three and adding the additional trace contaminants from materials and processes (Table 9), and from biological sources (Table 13). This is done in Table 17 to present a final generation rate pattern for the entire mission covering all sources of contamination.

The final step in this analysis requires that the generation rate patterns be translated into useful and practical programs for the operation of the TMCU.

It will be noted that no substitutions have been made for heptane, methane, and hydrogen although they are all much alike in chemical and physiological behavior. It is suggested that they all be used in the TMCU because substitutions for such large amounts of materials might lead to gross errors.

TABLE 17

FINAL GENERATION RATE PATTERN FOR MATERIALS, PROCESSES AND MEN (14-DAY TOTALS)

				moles moles
Summation Over itute Trace Materials	Amount, µ Moles	114,000 16,395 6,192 1,440 294 3,540	1,218,000 8,200 203	4,200,000 206,000 100,800 37,500 10,080 1,200 in Table 15.
Summat Substitute	Material	Ace tone CO** NBAL HEP BEN**	F114 EDC BEN**	4,200,000 NH3 206,000 CH4 100,800 H7 53,900 CO** 10,080 PH 1,200 UNK*** times the amounts listed total CO = 16,395 + 53
erial	Amount u Moles		1,218,000 8,120 33 47 47 203	4,200,000 206,000 100,800 53,900 10,080 1,200 times the at total CO =
Substitute Trace Material	"Relative Toxicity"	tronic Gear	 480 480	es are three in one source:
Substit	Name or Symbol	r than Blectronic	(F114) (EDC) EDC EDC EDC (BEN)	(NH ₂) (CH ₄) (CO) (PH) by more than
	"Relative Toxicity"	Sources Other	480 23 (est.) 37	electronic gear.
Actual Trace Material	Amount, u Moles	le and Process	1,218,000 8,120 406 406 203	Sources ,200,000 206,000 100,800 10,080 1,200 1,200 nts all
Acti	Symbol	SCS* (See Table 15) Material a	F114 EDC VDC MC BEN	NH3 4 CH4 H, CO PH UNK * Represe

^{***}The existence of these two groups of unknown contaminants from electronic gear and biological sources suggests the need for further analytical studies to identify these trace materials.

Programming the Trace Material Control Unit

The absolute generation rates presented in Table 16 must be integrated into programs for operation of the Trace Material Control Unit to simulate contamination of a system which does or does not involve a trace material sink. These programs are developed in the following sections of the report. Their development required (1) the use of a simple mathematical model simulating the depletion process, (2) the selection of a reasonable leak rate or circulation rate through the Environmental Control System (ECS), and (3) the definition of complete systems adapted for use with the TMCU.

Simulation of Depletion Processes

The possible depletion factors are gas sampling, condensation, decompression, operation of the Environmental Control System (ECS), and leakage. Gas sampling is analogous to removal of trace contaminants by the ECS or by leakage. If all three are 100 percent effective in the removal of trace materials, gas sampling can be included as an insignificant contribution to the other two depletion factors. Condensation is not expected to be a significant depletion factor providing there is good circulation coupled with an efficient ECS, as appears to be the case. Condensation would be a major depletion factor to a closed system containing no other sinks for trace materials. This does not appear to be a practical situation to simulate. The system could neither be controlled nor used to any purpose. Decompression is a special event which would be handled as a purging followed by a new approach to steady state conditions. Thus, ECS operation and leakage constitute the depletion factors treated for modification of generation rate patterns.

Leakage and ECS operation can be treated alike in that both remove trace contaminants at rates proportional to the concentration of contaminant. Assuming complete mixing in the cabin, the steady state concentration of any contaminant can be obtained, using appropriate units, from:

$$D_{ss} = r C_{ss}$$
 [2]

where r = leak rate or rate of circulation through the ECS,

 $D_{\alpha\alpha}$ = depletion rate at steady state.

The above condition is preceded by a transient or unsteady-state process leading to the steady-state concentration. This process can be treated as a simple dilution problem having a particular solution corresponding to the steady-state condition described above.

If R = rate of generation of trace material

Q = rate of generation of clean air as by

- (1) leakage and replacement of contaminated air, or
- (2) flow through the ECS, in cfm for example

v = volume of system

t = time

C = concentration,

then
$$\frac{dc}{dt} = \frac{R}{v} - \frac{QC}{v}$$
 or $\frac{vdc}{dt} + QC = R$

This homogeneous linear first order differential equation has a particular solution for the steady-state condition when $\frac{dc}{dt} = 0$:

$$R = Q C_{ss}$$

It has the general solution,
$$C = \frac{R}{Q} \left(1 - e^{-\frac{Qt}{V}} \right)$$
 [4]

Selection of Reasonable Leak Rate and Circulation Rate through Environmental Control System

An estimation of circulation rate through the Environmental Control System, ECS, is based upon a rate sufficient to maintain the cabin ${\rm CO}_2$ level at one percent, or at a partial pressure of 7.66 mm Hg. The uncontrolled accumulation of ${\rm CO}_2^{(5)}$ with time varies with cabin volume because of man's accomodation to ${\rm CO}_2$ and retention of ${\rm CO}_2$ in his system. In 1000 liters, the pressure of ${\rm CO}_2$ from one man reaches 36 mm Hg in three hours. This corresponds to an average generation rate of 200 liter-mm Hg per minute. For a volume of 2000 liters per man as in the case of the Apollo C/M, the generation rate is about 250 liter-mm Hg per man-minute. Under this condition, a circulation rate of 33 liters per minute through the ECS would be sufficient to maintain a steady state partial pressure of ${\rm CO}_2$ of 7.6 mm Hg.

The total circulation required for three men would be 99 liters per minute, assuming no other sources of CO₂, and assuming 100 percent efficiency of the ECS. This amounts to 3.5 cfm or 1.65 percent per minute. For purposes of this study, the value of 3.5 cfm is used either as a leak rate or rate of circulation through the ECS.

The time to reach steady state based on either leak rate or circulation rate can be calculated from the equation:

$$C = \frac{R}{Q} \left(1 - e^{-\frac{Qt}{v}} \right)$$

where C = "concentration" of CO₂ in mm Hg

 $R = \text{rate of generation of CO}_2 \approx 760 \text{ 1-mm Hg/min}$

Q = rate of generation of clean air ≈ 100 1/min

t = time in minutes

v = volume = 6,000 1

The time to arrive within one percent of the steady state concentration is 4.6 hours.

The units used in the above calculation are less convenient than:

 $C = \mu \text{ moles/cu ft}$

 $R = \mu \text{ moles/hour}$

Q = cu ft/hour = 60 x cfm = 60 x 3.5 = 210

v = cu ft of system = 212

t = hours

Noting that $Q \cong v$ or $Q/v \cong 1$, the equation takes the simpler form:

$$c = \frac{B}{Q} \left(1 - e^{-t} \right)$$
 [5]

A TMCU Program for Closed System without Trace Material Sinks

Imagine a closed system with zero dumping rate. The TMCU is to be programmed to simulate the constant rate of increase of trace material concentration in a closed system having no sinks for removal of trace contaminants. In this case, the pulse rate will be constant and equal numerically to the calculated generation rate in μ moles per hour.

Table 18 contains a compilation of generation rates for all trace materials (Column 3). It will be observed that only ammonia calls for a generation rate or pulse rate slightly exceeding the designed capability of the Trace Material Control Unit, 10,000 μ moles per hour or pulses per hour.

In view of the probability that no system will be free from condensation and adsorption, it is unlikely that the final concentration of Column 4 (Table 18) will be attained after 14 days of generation into a closed system. The error might, in fact, be extremely great in the cases of benzene, heptane, n-butyl alcohol, and phenol, which are generated at low rates and never reach high concentrations.

It is suggested, therefore, that the closed system be simulated by an open-loop system from which trace materials are dumped continually at a rate proportional to their concentration, i.e. at a constant-volume rate. The pulse rate from the TMCU will have to exceed the calculated rate of generation of trace materials by the rate of loss by dumping:

Pulse rate = generation rate + rate of loss by dumping.

The rate of loss by dumping will be directly proportional to concentration.

The concentration, in turn, must vary from zero at zero time to a final value at 336 hours of:

generation rate (in µ moles/hour) x 336 hours Volume (212 cu ft)

TABLE 18

CALCULATED CONCENTRATIONS OF TRACE MATERIALS

centration**	52.5	15.3	2.6	1.5	1.2	0.76	0.12	0.11	90.0	0.018	0.0063
Steady State Concentration** u Moles/Cu Ft	59.4	17.3	2.9	1.7	1.4	0.76	0.14	0.12	0.088	0.020	0.0071
Final Concentration with No Sinks*, u Moles/Cu Ft	19,800	5,740	970	538	475	254	, 47.5	38.7	29.2	6.8	2.3
Average Generation Rate, u Moles/Hour	12,500	3,620	613	340	300	160	30	24.4	18.4	4.3	1.5
Total Amount for 14-Day Mission, u Moles	4,200,000	1,218,000	206,000	114,000	100,800	53,900	10,080	8,200	6,190	1,440	497
Material	NH ₃	F114	GH ₄	Acetone	ਜ	8	Pheno1	EDC	NBAL	HEP	BEN

* For total output of generator into 212 cu ft.

**Same as above, but with either a leak rate of 3.5 cfm or circulation through 100 percent efficient ECS at 3.5 cfm.

Thus, the concentration is directly proportional to time. Accordingly:

Pulse rate = generation rate + bt = a + bt.

Taking Freon 114 as an example, its generation rate is 3620 μ moles per hour. From this, a = 3620 μ moles per hour. The final concentration at 336 hours will be (3620 x 336)/212 or 5740 μ moles per cu ft, at which time the rate of loss by dumping is:

 $Cr = 5740 \mu \text{ moles/cu ft x 3.5 cfm x 60 min/hr}$

= bt

. b = 3590, and

Pulse rate = a + bt = 3620 + 3590 t.

Table 17 lists the data necessary to calculate the constants a and b, and Table 19 outlines the program for the trace material control unit for all trace materials in terms of the constants a and b. The program is displayed graphically for BEN, HEP, NBAL, EDC, and PH in Figure 5. For these materials, the final pulse rates do not exceed the designed capabilities of the TMCU. This open-loop system will be useful for studies of these materials without much danger of failure due to adsorption or condensation.

If the simulation is to include all the trace materials at one time, the open-loop system will still be more reliable. To use it, the TMCU must be designed to put out from over four moles per hour of ammonia to only 1.5 u moles per hour of benzene.

PROGRAM FOR TMCU TO SIMULATE SYSTEM
HAVING NO TRACE MATERIAL SINKS*

<u>Material</u>	Initial Pulse Rate, a, or Actual Generation Rate in u Moles/Hr	Final Concentration y Moles/Cu Ft	$b = \frac{d(PR)^{**}}{dt}$ Pulses/Hr ²	Final Pulse Rate, Pulses/Hr
NH ₃	12,500	19,800	12,360	4,170,000
F114	3,620	5,740	3,590	1,210,000
CH ₄	613	972	606	205,000
Acetone	340	539	336	113,000
H ₂	300	476	297	100,260
со	160	254	158	53,500
PH	30	47.6	30	8,490
EDC	24.4	38.7	24.2	8,140
NBAL	18.4	29.2	18.2	6,114
HEP	4.3	6.82	4.2	1,440
B EN	1.5	2.38	1.5	500

^{*} Based on open-loop system having a dumping rate of 3.5 cfm and volume of 212 cu ft.

^{**}Equals (final concentration) (3.5/336).

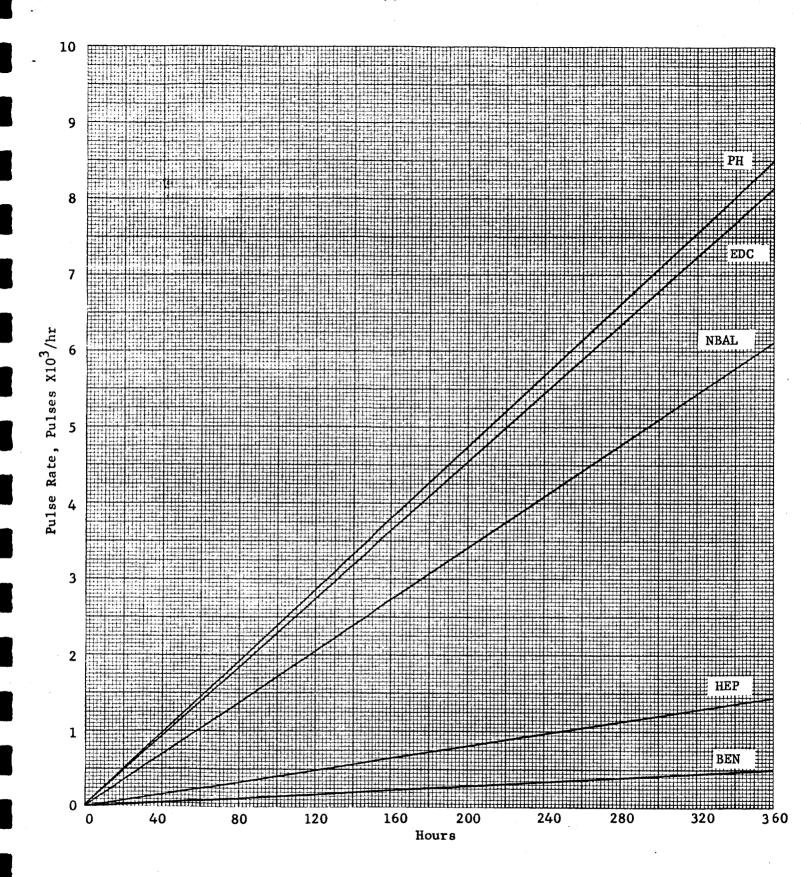


FIGURE 5 PROGRAM OF TMCU TO SIMULATE SYSTEM WITHOUT TRACE MATERIAL SINKS

Programming the TMCU to Simulate a Closed System with Trace Material Sinks

This simulation calls for reliably maintaining steady-state concentrations over a long period of time. This can be accomplished by introducing trace materials into a closed system during the transient period until the steady-state concentration is attained, and then dumping the system while continuing to introduce trace materials at a constant rate equal to the rate of dumping. The transient period for a 212 cu ft system is 4.6 hours when simulating a leak rate of 3.5 cfm or for the same rate of circulation through an efficient environmental control system. The relationship between concentration and time during this transient period was given earlier in the report as:

$$c = \frac{R}{Q} \left(1 - e^{-\frac{Qt}{V}} \right)$$

For purposes of this simulation,

 $C = concentration of trace contaminant in <math>\mu$ moles/cu ft

R = rate of generation in \u03c4 moles/hour

Q = rate of dumping in cu ft/hour = 60 x (3.5 cfm)

t = time in hours.

It follows that,

$$\frac{dm}{dt}$$
 = pulse rate = $v \frac{dc}{dt}$ = R - QC = R \in $\frac{Qt}{v}$

The volume, 212 cu ft, is, by chance, almost equal to Q. Therefore,

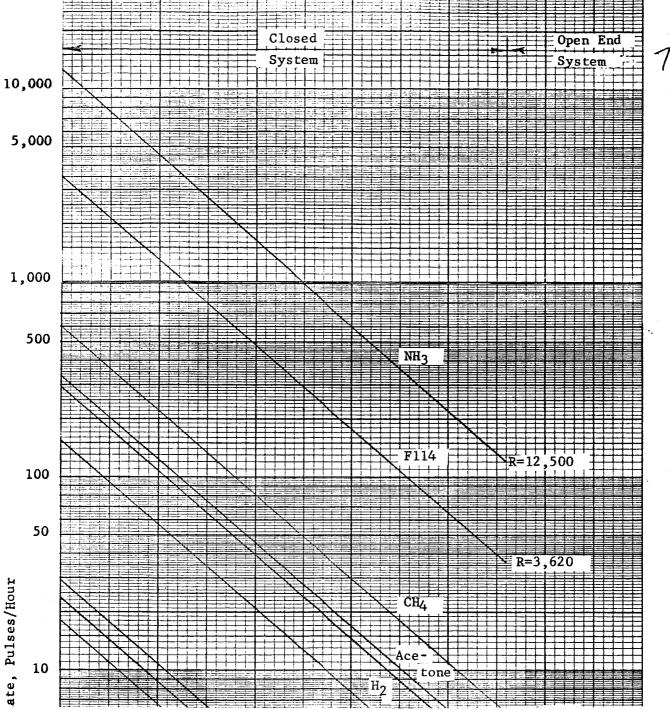
Pulse rate =
$$\frac{dm}{dt}$$
 = R e^{-t} .

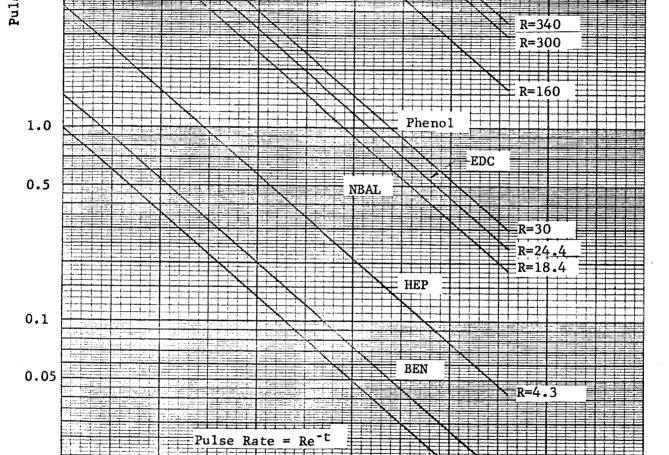
The pulse rate during the first 4.6 hours is a multiple of e^{-t}, the coefficients of e^{-t} being the generation rates of Column 3, Table 18. The pulse rates after 4.6 hours will vary from material to material as:

C (steady state) =
$$\frac{R}{Q}$$
 = $\frac{Pulse\ Rate}{Q}$

The steady-state concentrations are given in Table 18. The pulse rate during the transient and steady-state period is presented graphically in Figure 6.

Again, ammonia is the only material to exceed the capabilities of the TMCU (but by only 25 percent). The high generation rate for ammonia may not be realistic. It is a maximum rate obtained by assuming neglect of usual hygiene, and consequent bacterial degradation of the urea content of sweat.





5

FIGURE 6 PROGRAM FOR TMCU TO SIMULATE LEAK OR OPERATING ECS

Time, Hours

73

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